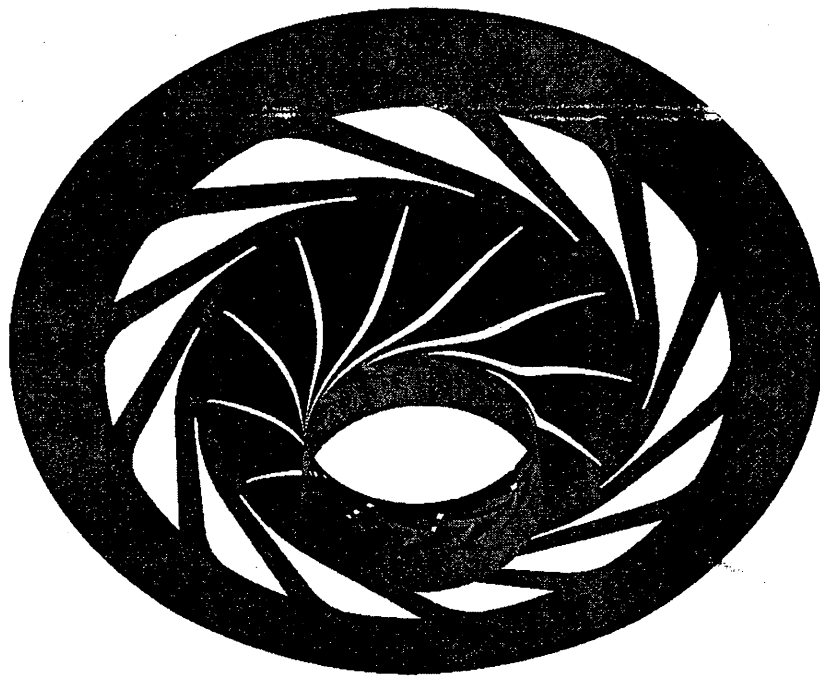




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**Steady and Unsteady Simulations of the Flow in an  
Impeller/Diffuser Stage**



**F. Canabal, D. Dorney, R. Garcia  
NASA Marshall Space Flight Center  
Applied Fluid Dynamics Analysis Group**



# Overview

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- Problem definition
- CFD for design
- Impeller/diffuser test case
  - TASCflow
  - Corsair
- Conclusions and future plans



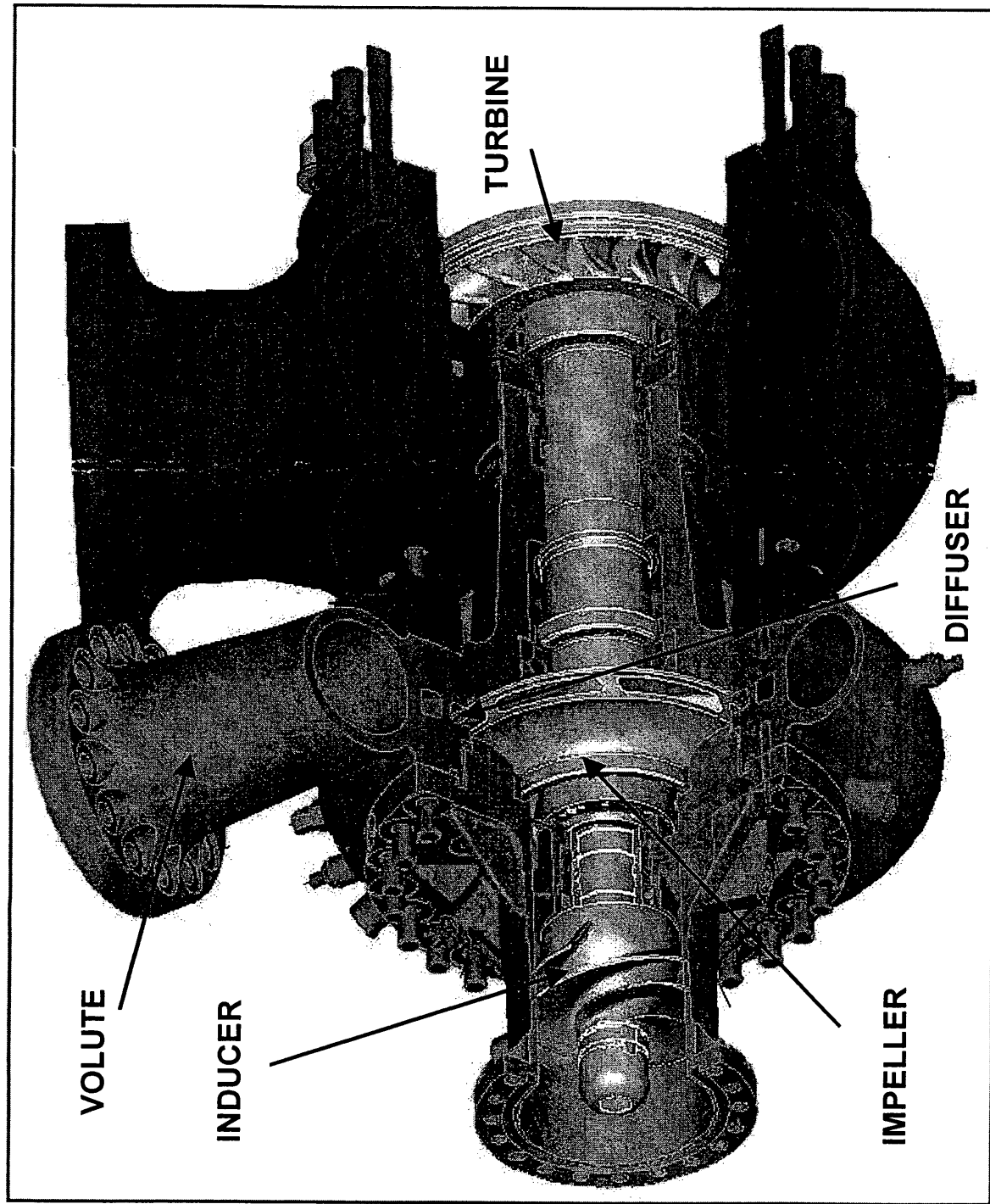
## Problem Definition

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- **SLI engine designs will require pumps to throttle over a wide flow range while maintaining high performance**
- **Unsteadiness generated by impeller/diffuser interaction is one of the major factors affecting off-design performance**



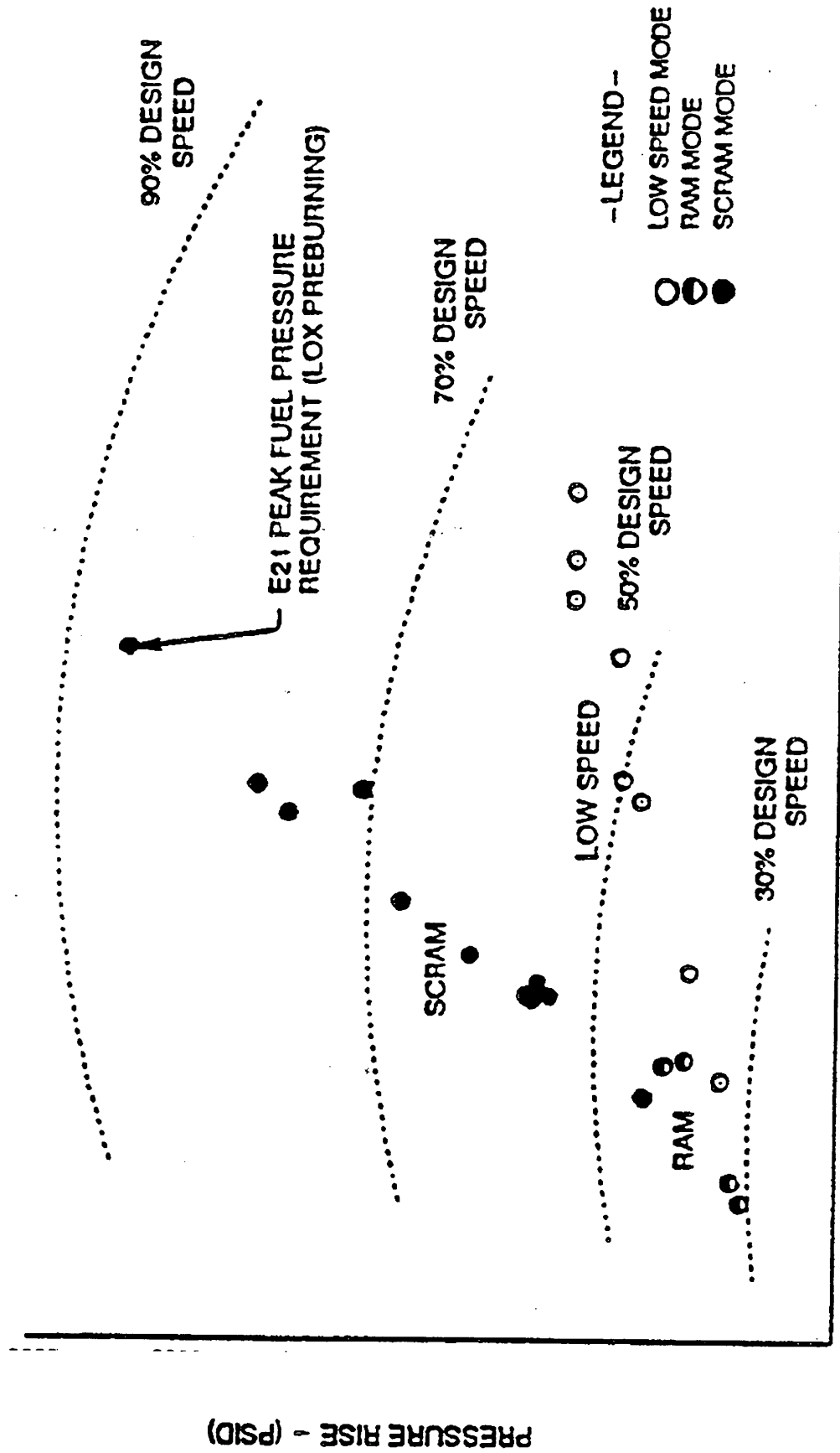
# Generic Oxidizer Turbopump



3/18/02



# Deep Throttling of Turbopumps



PUMP DELIVERED FLOWRATE (LBS/SEC)



# CFD for Design

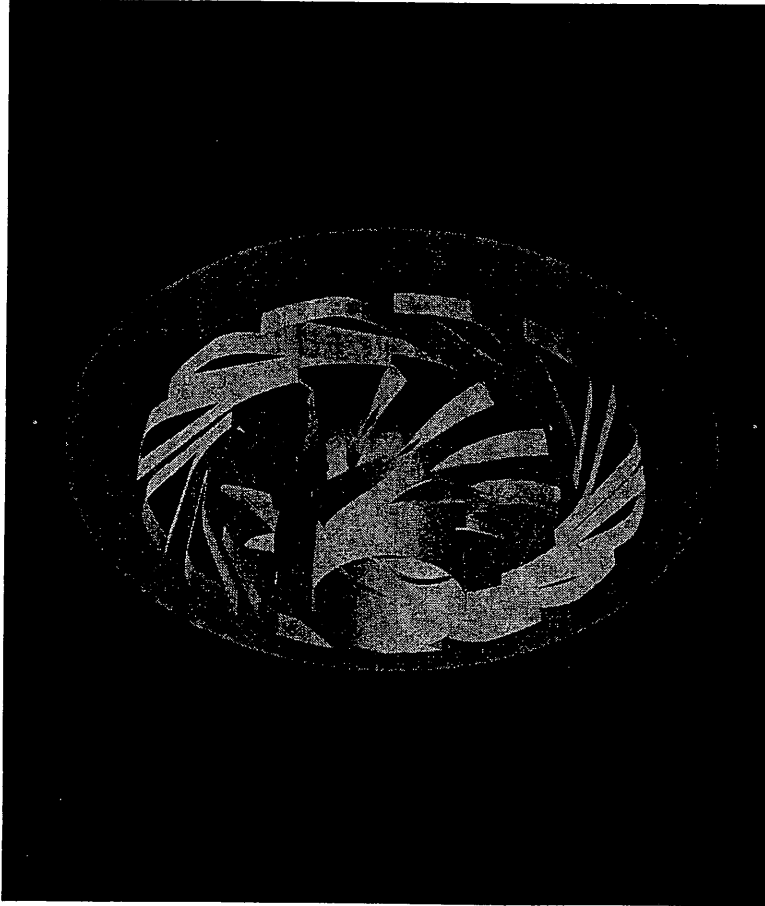
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- **Unsteady flow simulations necessary for predicting impeller/diffuser interaction**
  - Steady simulations miss important off-design physics
- **Grid density based on a compromise between computational efficiency and solution accuracy**
  - design parametrics require many simulations in a short time frame
  - Simulations must resolve pertinent flow physics and unsteady frequencies (separation, stall cells, vortex shedding, etc.)



# Pump Stage Test Case

## HIGH-HEAD IMPELLER AND VANE ISLAND DIFFUSERS



IMPELLER BLADES (FULL)	6
IMPELLER BLADES (PARTIAL)	6
DIFFUSER VANES	13
FLUID	WATER
FLOW RATE	1210 GAL/MIN
ROTATION RATE	6322 RPM
FLOW COEFF.	0.144
TIP SPEED	249 FT/SEC
INLET SWIRL	63 DEG



# CFX/TASCflow

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- **Pressure-Based Coupled Solver**
- **Conservative Finite Element-Based Control Volume Method**
- **Advection Modeling Uses Skew Upstream Differencing Schemes with Physical Advection Correction**
- **Steady/Transient Analysis**
- **Compressible/Incompressible**
- **Stage/Frozen Rotor/Transient Rotor-Stator Interaction**
- **Moving Grid**
- **Parallel Implementation with PVM**



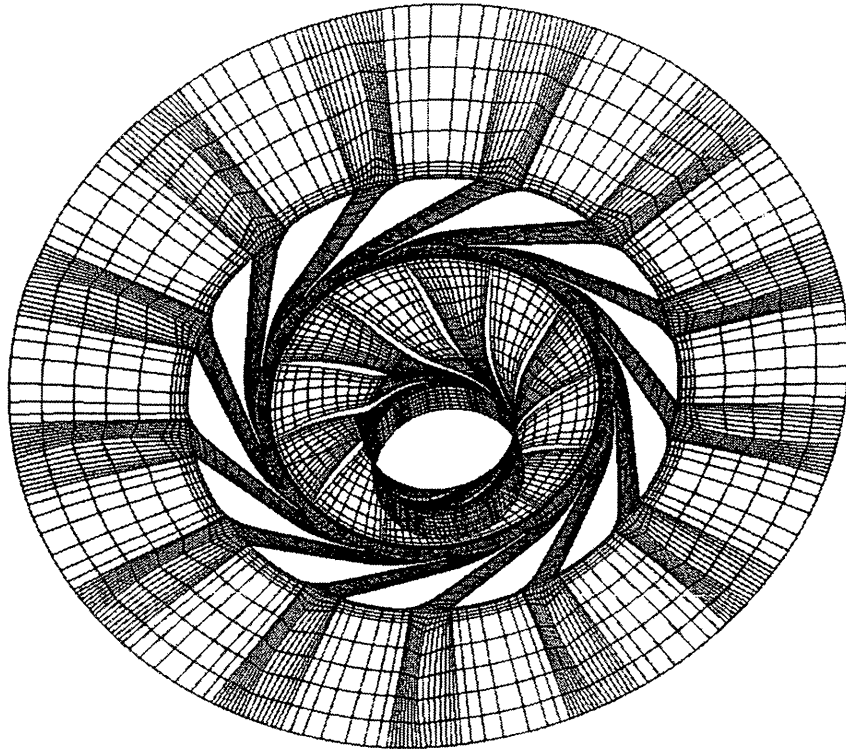


# CFX/TASCflow Simulation

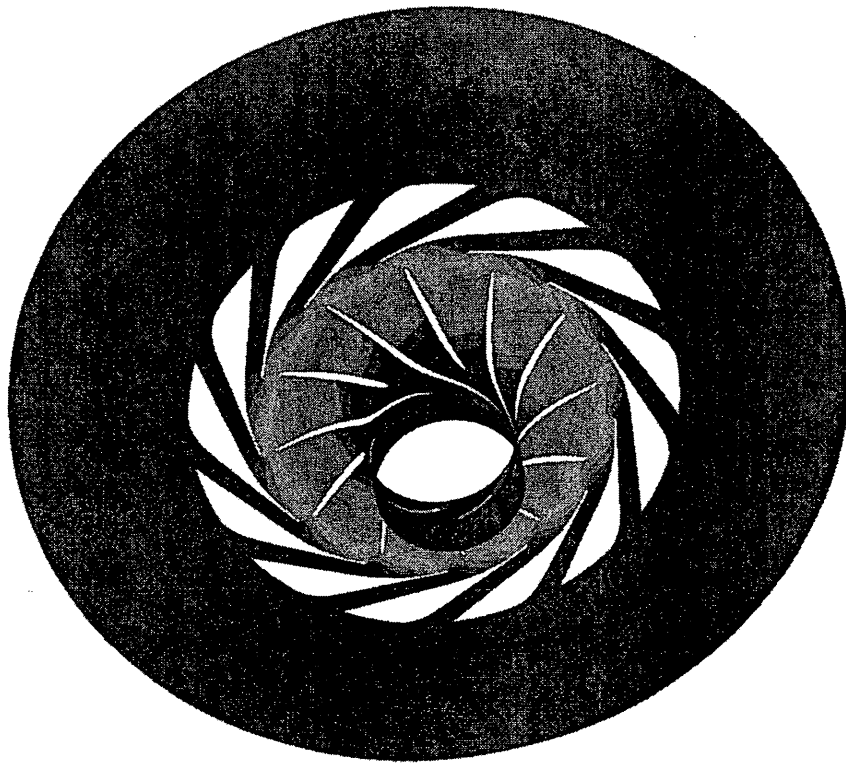
**6 FULL + 6 SPLITTER IMPELLER BLADES AND 13 DIFFUSER**

**VANES - 100% DESIGN FLOW - 300,000 GRID POINTS**

**RUN TIME: 1,152 cpu hours/revolution – 6 R12000 300 MHz processors**



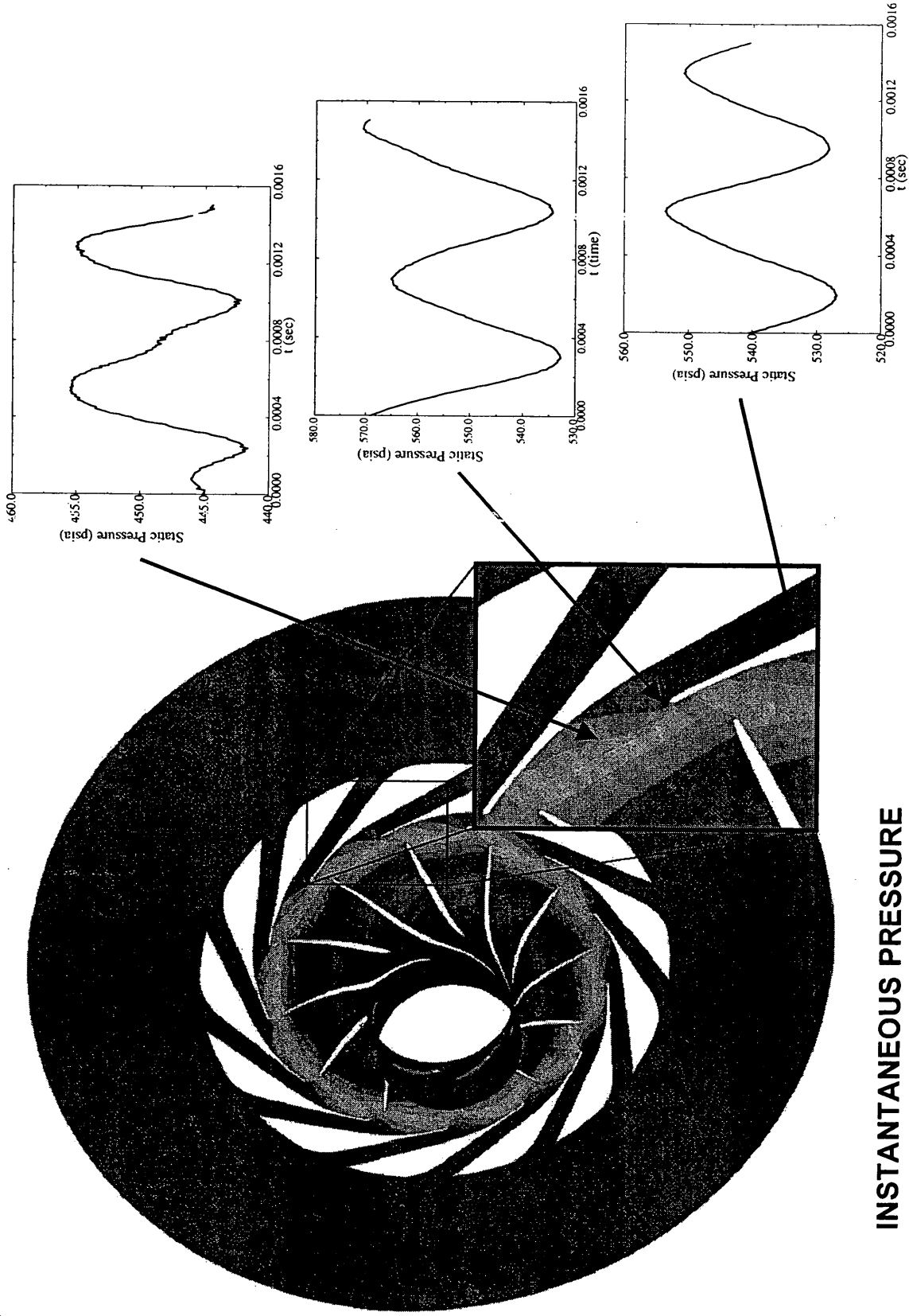
**COMPUTATIONAL GRID**



**INSTANTANEOUS PRESSURE**



# Unsteady Pressure

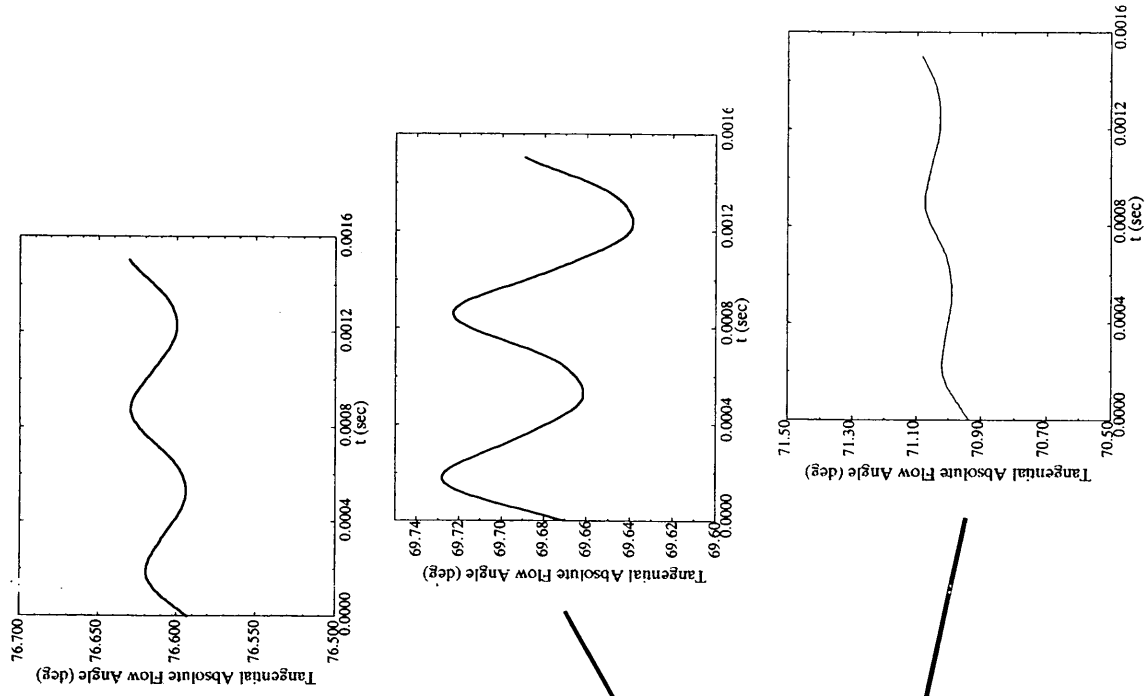
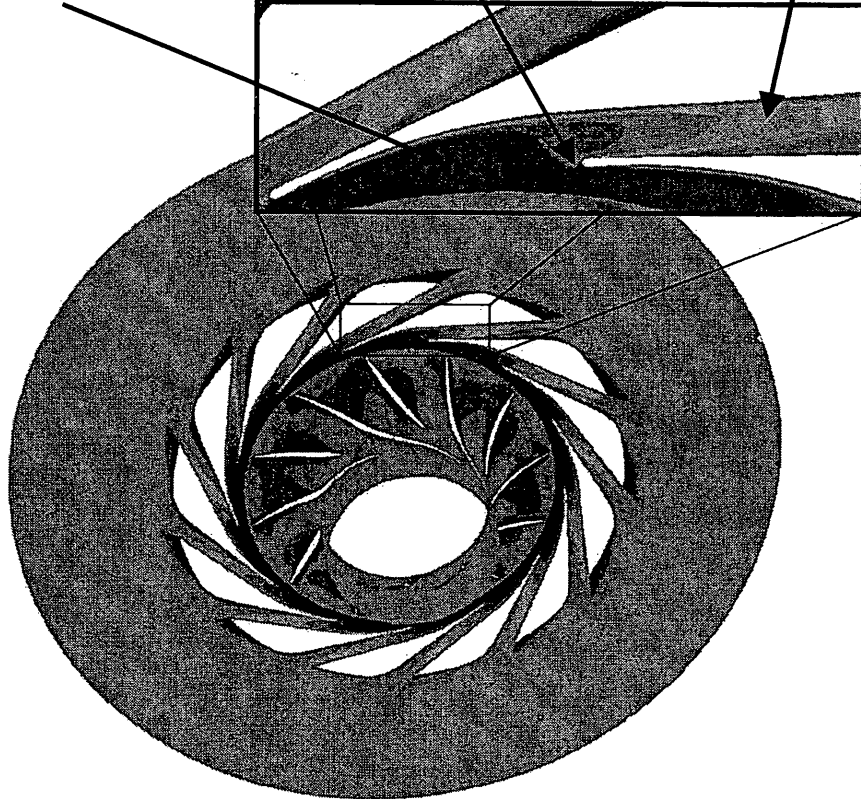


INSTANTANEOUS PRESSURE



# Unsteady Flow Angle

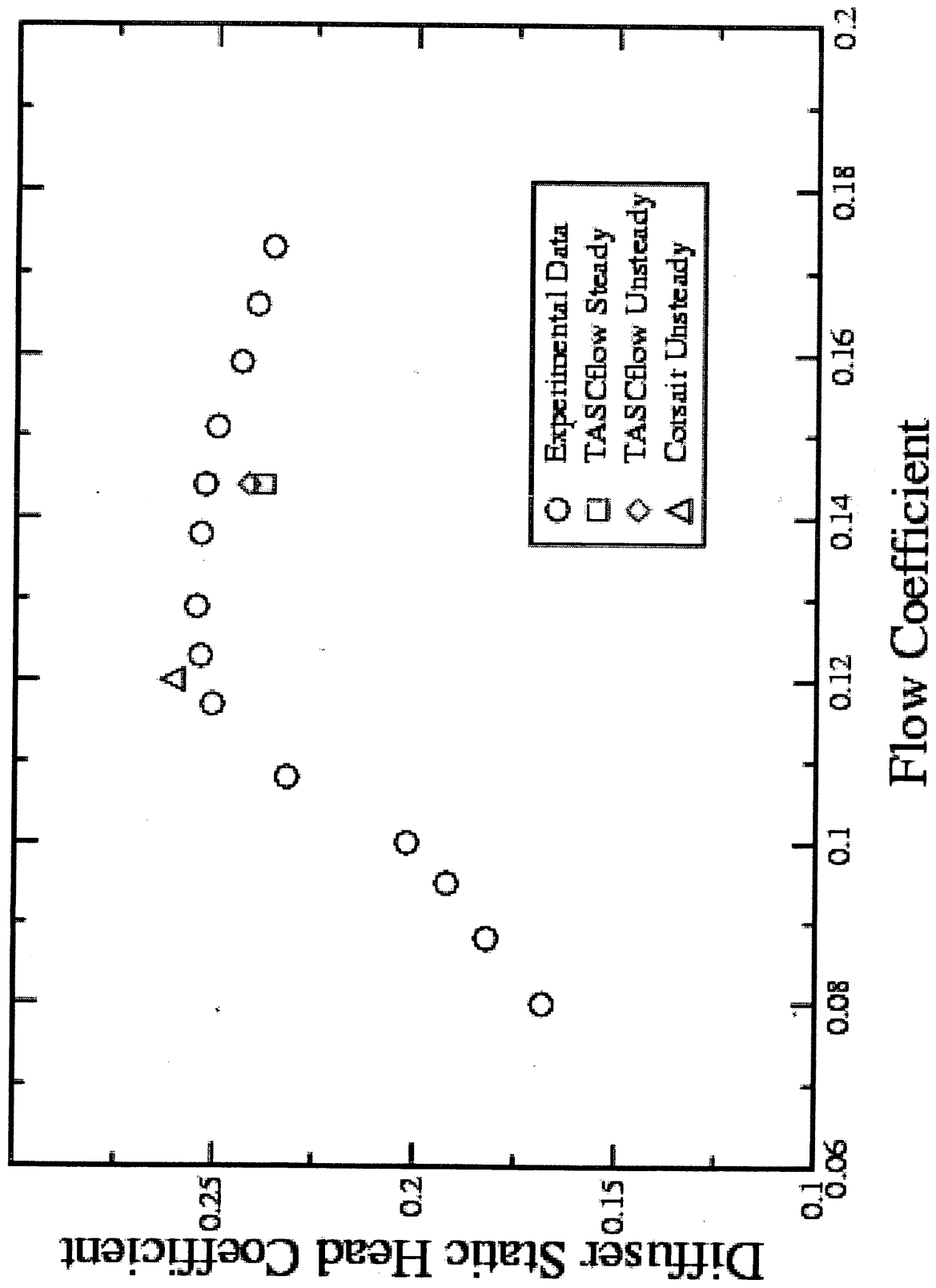
SPEED  
2.805E-03  
2.665E-03  
2.525E-03  
2.384E-03  
2.244E-03  
2.104E-03  
1.964E-03  
1.823E-03  
1.683E-03  
1.543E-03  
1.402E-03  
1.262E-03  
1.122E-03  
9.820E-02  
8.417E-02  
7.014E-02  
5.611E-02  
4.208E-02  
2.805E-02  
1.402E-02  
0.000E+00



INSTANTANEOUS VELOCITY MAGNITUDE



# Time-Avg Diffuser Static Head Coefficient





# CORSAIR

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- Time-dependent equations of motion
- Implicit solution scheme
- Third-order spatial discretization of inviscid fluxes
- Second-order spatial discretization of viscous fluxes
- Second-order temporal accuracy
- Multi-block O-H grid topology
- Grid movement simulations blade motion
- MPI and OpenMp



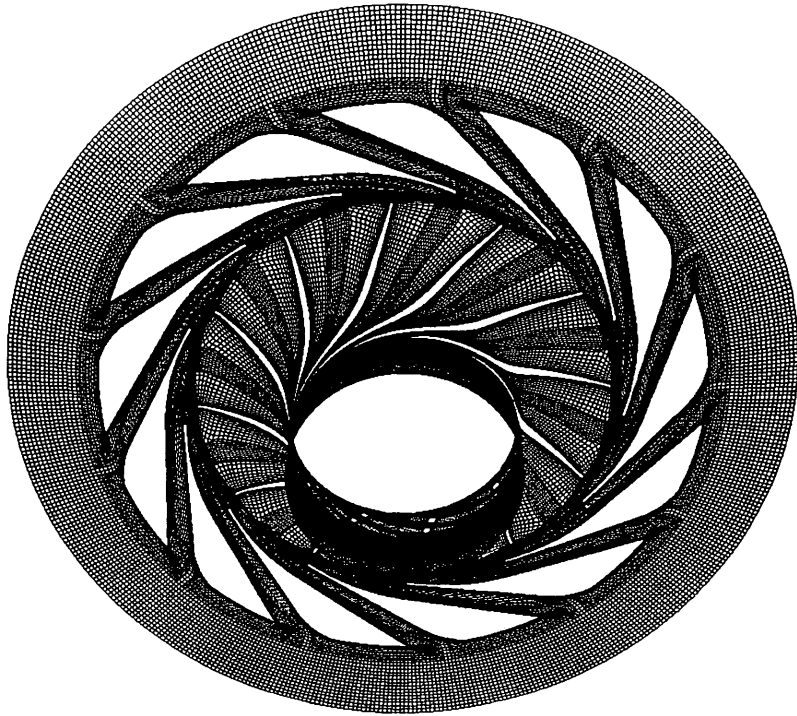
# Corsair Simulation

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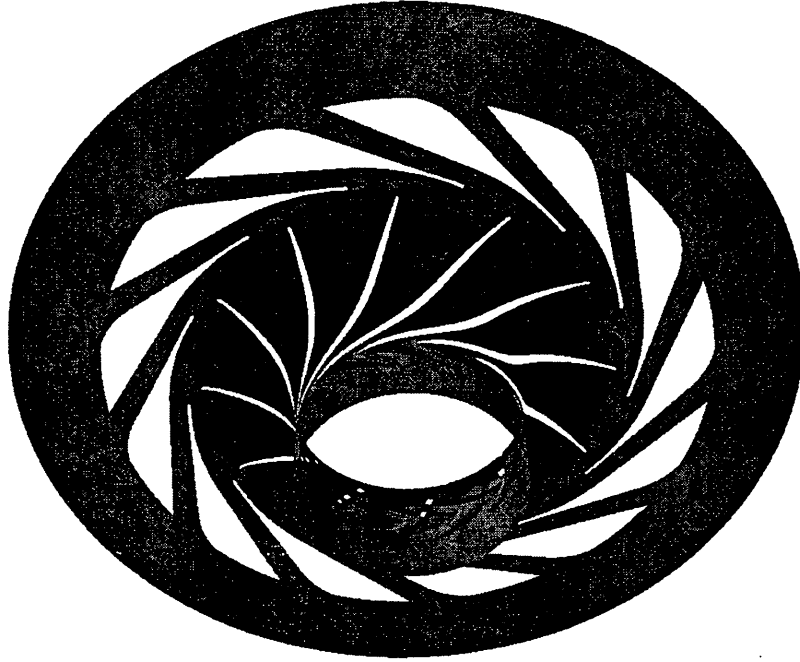
**12 FULL IMPELLER BLADES AND 13 DIFFUSER VANES**

**83% DESIGN FLOW - 4.5 MILLION GRID POINTS**

**RUN TIME: 250 cpu hours/revolution – 32 R12000 400 MHz processors**



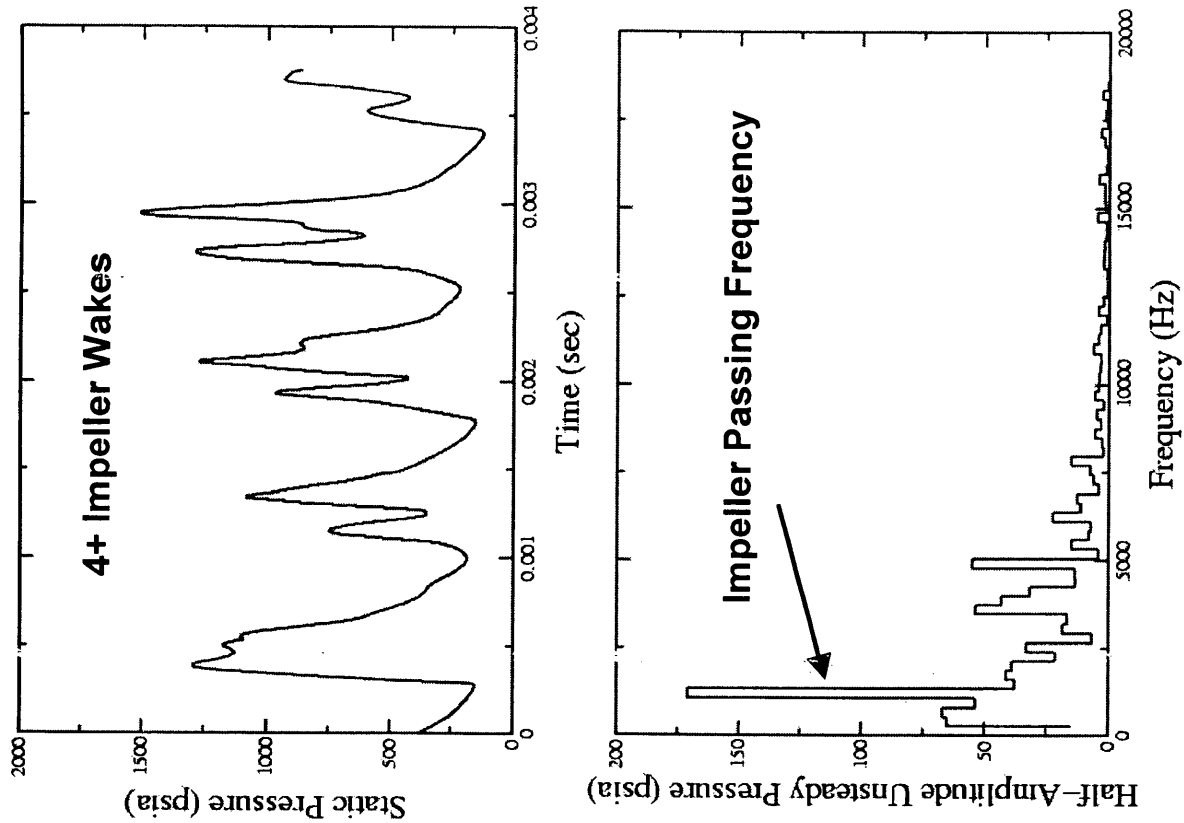
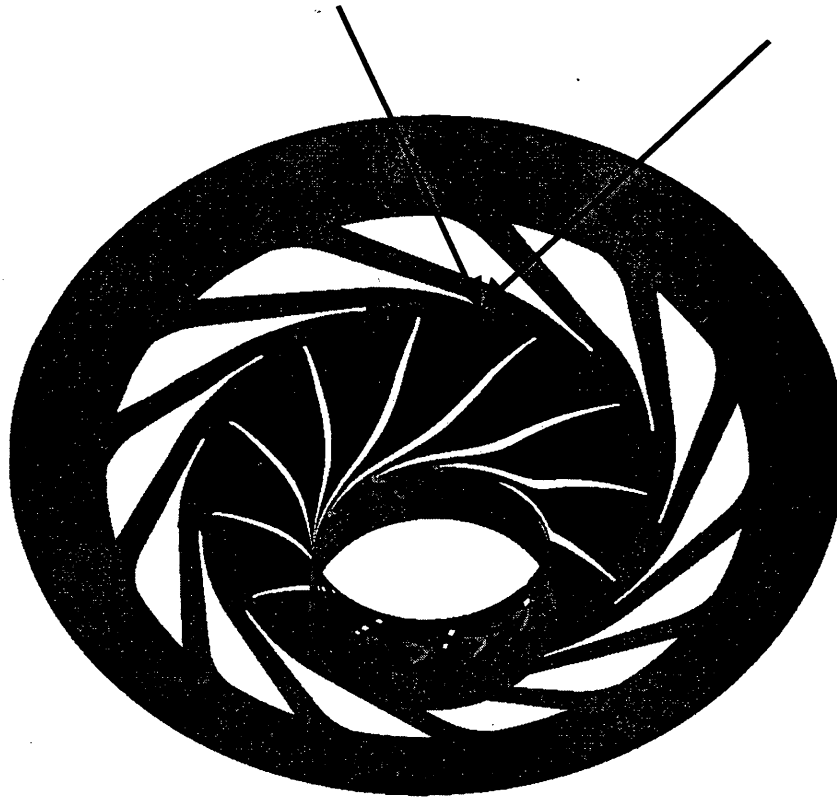
**COMPUTATIONAL GRID**



**INSTANTANEOUS PRESSURE**



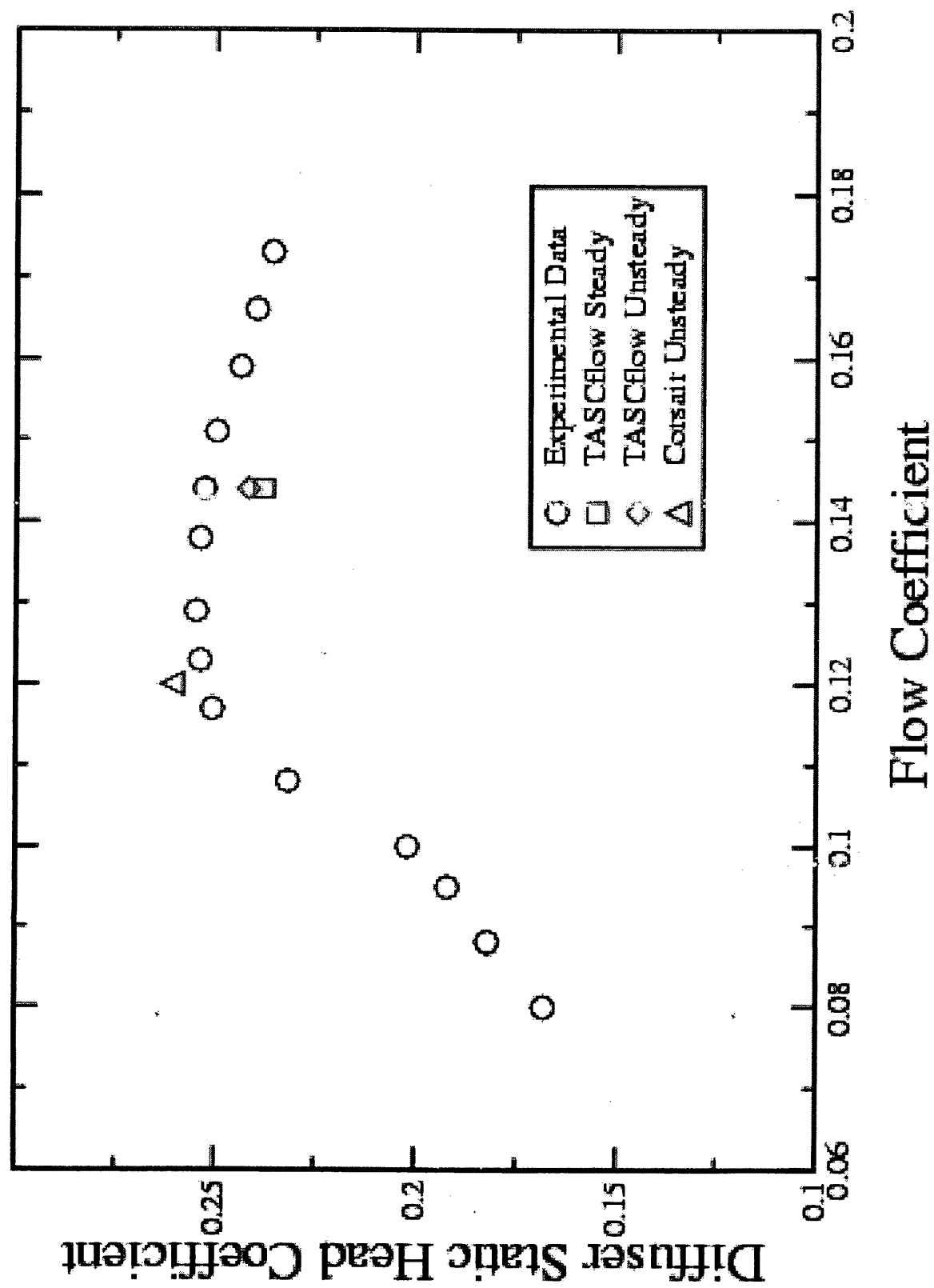
# Unsteady Pressure



INSTANTANEOUS PRESSURE



## Time-Avg Diffuser Static Head Coefficient

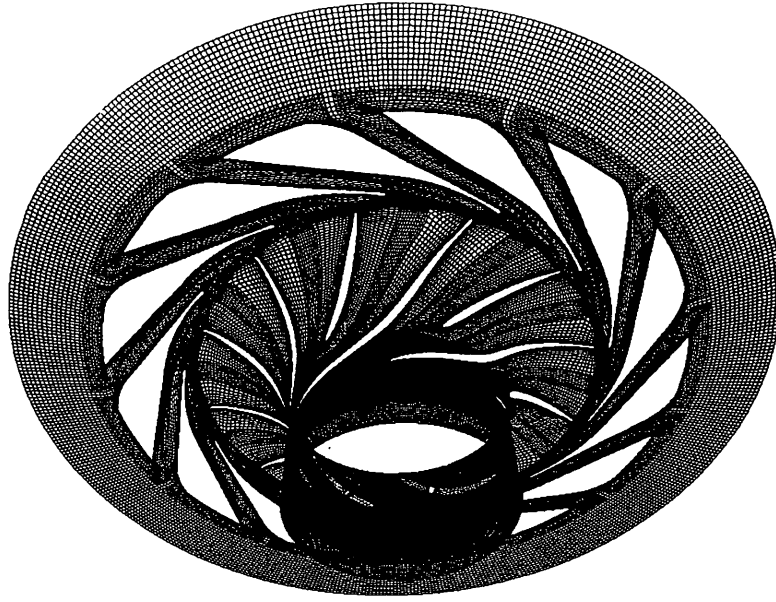




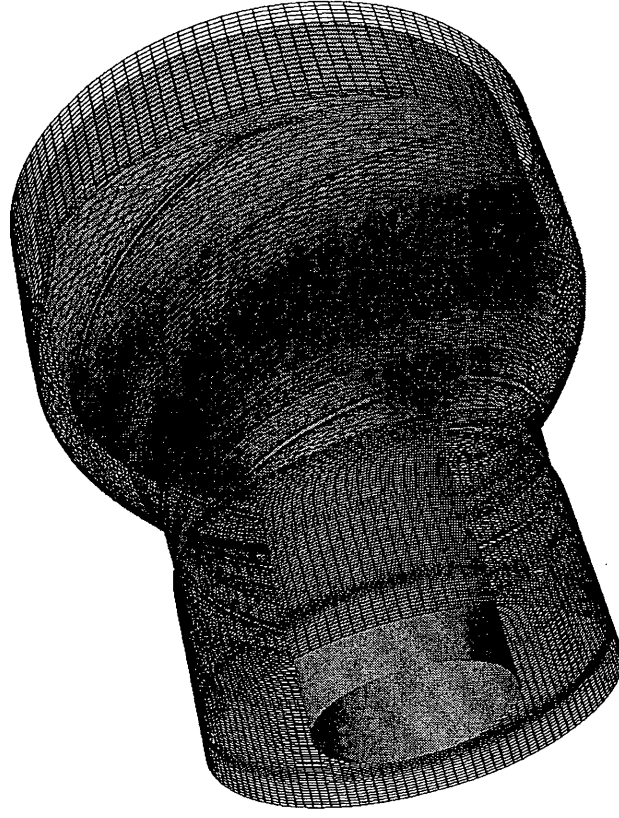


# Pumps and Inducers

Full-Annulus Simulations in Progress



IMPELLER W/SPLITTERS



INDUCER



# Inducer Instantaneous Pressure Field

Dimensional Pressure in psia = Non-Dimensional Pressure x 21.9





## Conclusions and Future Plans

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- Initial unsteady simulations completed for impeller/diffuser stage
- Corsair simulations to continue across wide flow range and for inducer/impeller/diffuser combination
- Results of unsteady simulations being used to guide/explore new designs